



#### NATIONAL AERONAUTICS AND SPACE ADMINISTRATION



# TRANSLUNAR, LUNAR PARKING ORBIT, AND TRANSEARTH PROCEDURES

C-PRIME MISSION

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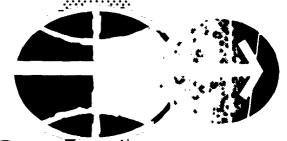
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MANNED SPACECRAFT CENTER HOUSTON, TEXAS

# TRANSLUNAR, LUNAR PARKING ORBIT, AND TRANSEARTH PROCEDURES

C-PRIME MISSION

AS-503/CSM-103 12 November 1968

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#### 1.0 PURPOSE

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This document contains the nominal primary crew procedures for the CSM-103 spacecraft during the lunar Mission C-Prime. The specific time period covered by these procedures is from CSM/S-IVB separation at 3:20 GET through the last TEMCC (MCC7) at 144:50 GET.

The purpose of the Translunar, Lunar Parking Orbit and Transearth procedures document is to provide a single source of guidance, navigation and control procedures information for use in flight planning, in crew training, and in preparation of onboard checklists.

This document is a control document for this phase of the Mission C-Prime nominal crew procedures. Revisions to this document are subject to approval by the Procedures Configuration Control Board. Comments should be directed to Mr. Duane K Mosel, Flight Procedures Branch, Extension 5340 or to Stephen G. Paddock, Jr., Apollo Flight Crew Support Group. Houston Operations, McDonnell Douglas Astronautics Company, Extension 6101.

## 2.0 LIST OF ACRONYMS AND ABBREVIATIONS

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AOS	Acquisition of Signal
ATT	Attitude
BEF	Blunt End Forward
CB	Circuit Breaker
CDR	Commander
CM	Command Module
CMC	Command Module Computer
CMP	Command Module Pilot
COAS	Crew Optical Alignment Sight
CSM	Command and Service Module
DAP	Digital Autopilot
DB	Deadband
DSKY	Display and Keyboard
DV	Delta Velocity
EMS	Entry Monitor System
ET	Event Timer
FDAI	Flight Director Attitude Indicator
FPS	Feet Per Second
GDC	Gyro Display Coupler
GETI	Ground Elapsed Time of Ignition
GMBL	Gimbal
GND	Ground
GPI	Gimbal Position Indicator
нА	Apogee Altitude
HOR	Horizon
HP	Perigee Altitude
IMU	Inertial Measurement Unit
LEB	Lower Equipment Bay

LOS Loss of Signal
LM Lunar Module
LMK Landmark

LMP Lunar Module Pilot
LUI Lunar Orbit Injection

LV Launch Vehicle

MDC Main Display Console MGA Middle Gimbal Angle

MTVC Manual Thrust Vector Control

OPT Optics

ORDEAL Orbital Rate Drive Earth and Lunar

OSS Optical Subsystem

PAD Data Voiced to Crew From Ground

PB Pushbutton

PGNCS Primary Guidance, Navigation, and Control System

PIPA Pulse Integrating Pendulous Accelerometers

R Range Range Rate

REFSMMAT Reference Stable Member Matrix

RCS Reaction Control System
RHC Rotation Hand Controller

S-IVB Saturn S-IVB Stage

SCS Stabilization and Control System

SCT Scanning Telescope

SECS Sequence Events Control System

SEF Small End Forward

SEP Separation
SM Service Module

SPS Service Propulsion System

SC Spacecraft

S/U Setup SXT Sextant

TEMCC Transearth Midcourse Correction

TFI Time From Ignition

THC Translation Hand Controller

THETA Angle Between SC +X Axis and Local Horizontal

TIGN Time of Ignition
TLI Translunar Injection
TEI Transearth Injection

TLM Telemetry

TLMCC Translunar Midcourse Correction

TRUN Trunnion

TVC Thrust Vector Confrol VG Velocity to be Gained

(XX:XX) Inidcates ground elapsed time from liftoff in

hours: minutes

#### 3.0 NOMINAL MISSION PROCEDURES

The nominal Mission C-Prime Translunar, Lunar Parking Orbit, and Transearth flight crew procedures presented in this document are divided into two sections, a major activities timeline and standard procedures. The timeline identifies all guidance, navigation, and control related activities as a function of GET. The procedures necessary to successfully perform an activity required only once during the mission, e.g. CSM/S-IVB separation, are presented in detail within the timeline. Whereas, the procedures for an activity reoccurring periodically throughout the mission, e.g. an IMU alignment, are presented in detail in an addendum to avoid repetition. The timeline identifies how and when the standard procedure should be applied and the standard procedure is sufficiently flexible to cover any option required during the mission.

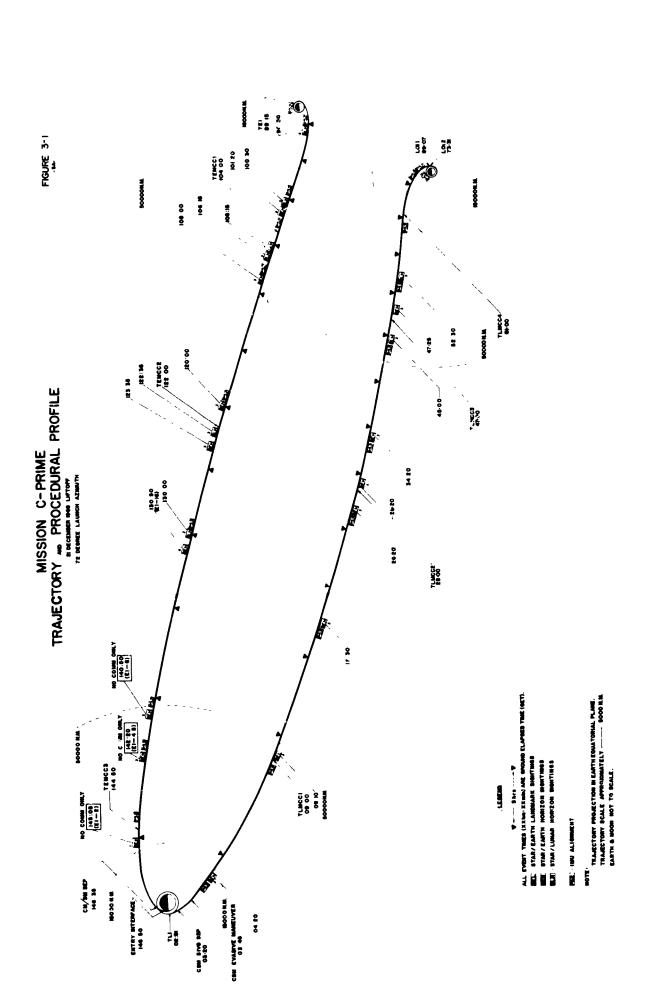
#### 3.1 <u>Major Activities</u>

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#### 3.1.1 Introduction

The major procedures timeline for the Mission C-Prime includes activities from after the TLI (2:51 GET) through the last Transearth midcourse correction (144:50 GET). Figure 3-1 shows the locations in time and approximate positions in space of selected major activities during the Translunar and Transearth phases. Figure 3-2 is a CSM attitude profile for the Lunar Parking Orbit phase. The nominal Mission C-Prime timeline assumes a 21 December 1968 launch, a 72 degree launch azimuth, and a nominal TLI ignition time of 02:50:31 GET.

The procedures presented in this document include primarily quidance, navigation, and control functions. Management of other systems, such as electrical power and environmental control are recommended as identified in Reference (1).



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			ATT DELAIN
c +		<b>8</b> 00	TRANS CONT PWR.ON (UP) RHC PWR OIRECT (BOTH) -MNA/MNR 1
- v a			SC CONTECTO CMC MONETRRE GMAG MODE(3)-RATER
_ α _ •	: <b>Z</b> li		AUTO RCS SEL(16)=MNA 8
> v	PROCEUJRES		VERIFY CH RCS LOGIC(2) -CLOSEDB TVC SERVO PWR 1-AC1/MNA
(00+00)	LTFT OFF		EDS PWR.ON(UP) VERIFY LV ENG LT-ON 1 FDAI/SW(ROTH) (ORDEAL) -INER 13
(02+51)	TRANSLUNAR INJECTION	-2MIN COR	START ET COUNT UP THC-ARMED
	ASSUMED PRE-CSM/S-1VB SEPAPATION SMITCH POSITIONS AND DAP CONFIGURATION (ADDENDUM 1)	•	RHC-ARMED CB PYRO A SEG A-CLOSED CB PYRO A SEG B-CLOSED CB SFCS ARM(ROTH)-CLOSED B SECS LOGIC (BOTH)-ON (UP)
(03+07)	CMP LOAD N17 WITH GEP ATTITUDE KEY V63E COR MONITOR S-TVH MANFUVER TO INERTIAL SEP ATTITUDE	Q Σ	PREPORT LOGIC ARM TO GNO RECEIVE GO FROM GNO NONESS BUS-MNA FLT RCDR-RECORD TLM INPUTS PCM-HIGH UP TLM CMD-RESET THEN NORM
(03+14)			COR PCM-PCM/ANL

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CB EDS(ALL)=OPEN CB PYRO A SEG A=OPEN CB PYRO R SEG B=OPEN 250	FLT PCDR-OFF(CTR) TAPE RCDR FWD-OFF(CTR) NONESS RUS-OFF	101	PR0 Key 00e	CMC MODE AUTO HMAG MODE (3) -RATE2  MAN ATT (3) -RATE CMD	OF NO	180 DEG PITCH AND 60 DEG Roll Left att maneuver Pro	KEY V49E 22 (COMMANDED R.P.Y) LOAD COMPUTED R.P.Y KEY V62E	PRO (COMMANDED R.P.V.) PRO (COMMANDED R.P.V.) MONITOR ATT MANEUVER OBSERVE S-IVB IN CMD WINDOW	
	ď ¥	8 0 0	37 BB	<b>G</b> 00	16 20		06 22	50 16 06 16	
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TAPE RCDR RCD-RCD TAPE RCDR FWD-FWD TAPE MOTION TB-RP	CMP COPY POST ILI NOS PROM DS57 KEY V32E VERIFY OSKY NVS ZERO	COR SECS PYRO ARM(ROTH)-ON(UP, B CM RCS LOGIC-ON(UP) 1 RCS CMD-ON 2 CMP KFY V60E (N174N20)	COR EMS FUNCTION-DV 1 FMS MODE-AUTO CMC MODE-HOLD	THC-THRUST(+)X AND HOLD (PRESEP TENSION)	CSM/LV SEP PB-PUSH AND HO, D 1 VERIFY LV ENG 1 LT-OFF 1 CSM/LV SEP PB-RELEASE 1	READ	A T T T T T T T T T T T T T T T T T T T	TVC SERVO PWR 1-OFF SECS LOGIC(BOTH) -OFF 8 SECS PYRO ARM(ROTH) -SAFE 8 CR SECS ARM(BOTH) -OPEN 8 CR SECS LOGIC(ROTH) -OPEN 8	•
		Z ¥		-2SEC	(03+20) •08EC	+5SEC			

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ATT D9-MIN RATE-LOW MAN ATT(3)-RATE CMD HOLD LOCAL VERTICAL ATTITUDE (DSKY THETA APPROX 270 DEG)	AND EARTH PRO KEY VATEATE	BMAG MODE(3)-ATT1/RATE2 EMS FUNCTION-DV EMS MODE-AUTO	THC-THRUST AFT (-X) UNTIL OSKY READS 1.5FPS VERIFY S-IVB RANGE OPENING VISHALLY THC-LOCKED	PRO TEX	KEY V66E (TRANSFER CSM SV F40M ČSM SLOTS TO LM SLOTS)	KEY V48E (DAP CONFIGURATION) LOAD 11112 (MAX D8.LOW RATE) 11111
	ž .	2	(94+35)	C4P F 37 88 P00		9 <b>9</b>
F 50 19 (COMMANDEU R.P.Y) KFY ENTER KEY V48E F 04 46 (DAP CONFIGURATION)	LOAD 11102 (MIN DR,LOW RATE) 11111 B	F J6 47 (CSM AND LM WT) VERIFY PAU DATA PHO F 06 47 (SPS GMBL TRIM)	COR PERFORM FORMATION FLYING AT 50 TO 100 FT TAKE PHOTOGRAPHS OF S-IVR	(03+30) PROCEDURES FOR CSM FVASIVE MANFIVER	Š	F 16 54 (N. KDOT. ITELA) CDR MAN ATT(3) -ACC CMD MANEUVER SC TO LOCAL VERTICAL (+X AXIS TOWARD EARTH) WHILE TRANSLATING TO MAINTAIN S-IVR IN CMD WILDOW

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NC. 1'(SIRIUS) (FAR HOR) NO. 15(PROCYON)(FAR HOR) NO. 15(PROCYON)(FAR HOR)	(05+00) LMP RECEIVE BLOCK DATA PAD From GND	(04.20) PROCEDURES FOR LOI POSITION DETERMINATION (ADDENDUM 5) (ADDENDUM 5)	(05.40) PROCEDURES FOR HI GAIN DWR UP POO LMP C8 HI GAIN ANT	GRP 2-CLOSED  223  HI GAIN ANT TRACK-MAN  RI GAIN ANT SERVO ELEC-PRIM  HI GAIN ANT BEAM-WIDE  HI GAIN ANT BEAM-WIDE	C.	
DRO PRO TE STATE OF THE STATE O	90	CDR MONITOR SC INERTIAL ATTIT::UE HOLD !!NTIL APROX (04+00)	PROCEDURES FOR IMU REALIGN TO REFSHMAT (ADDENDUM 2) RACKUP ALTAN NAV STADS	'n	(04+15) PROCEDURES FOR GDC ALIGN TO ILU (ADDENDUM 3)	PROCEDURES FOR STAR/EARTH HORIZON SIGHTINGS - 5 SETS OF 3 MARKS EACH (ADDENDUM 4) RECOMMENDED NAV STARS NO• 14 (CANOPUS) (FAR HOR) NO• 15 (SIRIUS) (FAR HOR)

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AEY V89E 04 06 (TRACK AXIS OPTION) 1040 00002 IN B2	18 (COMPUTED R.P. COPY R.P. Y	/37EOOE		**************************************	USE SPS FOR DV GREATER THAN SFPS TRIM DVS TO ZERO	PROCEDURES FOR SM RCS PROPULSION	(ADDENDUM 10) PROCEDURES FOR SPS THRUST SETUP (ADDENDUM 11)
is.	<b>1</b> 4	000	(08+20)			(0A+25)	(08+30)
HI GAIN ANT REAM=AS REGUIJED PERFORM HI GAIN ANT CHECK-UT	PROCEDURES FOR GDC DRIFT CHECV (ADDENDUM 7)	PROCEDURES FOR IMU REALIGN TO Refsmmat for drift check (addennum 2)	RACKUP ALIGN NAV STARS NO. 14 (CANOPUS) NO. 12 (RIGEL) LACKUP ALIGN CHECK NAV STAR NO. 15 (SIRIUS)	PROCEDURES FOR GOC ALIGN TO I JU (ADDENDUM 3)	PROCEDURES FOR GND UPLINK OF LSM STATE VECTOR AND P30 TARGET 10AD (ADDENDUM A)	FOLLOWING GND UPLINK LEB WHEN COMP LT-OUT (NO INTEG)	πυ -4
	(01+10)	(07+0)		(07+55)	(00+90)	000	9 L

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(00+60)	FIRST THANSLUNAR MIDCOURSE	(17.05) PROCEDUR	PROCEDURES FOR TERMINATING PTC (ADDENDUM 14)	
	PROCEDURES FOR SM RCS AND SPS PROPULSION MONITOR CHECK (ADDENDUM 10)	(17+10) PROCEDUR TO REF	PROCEDURES FOR IMU REALIGN TO REFSHMAT	5
(00+10)	PROCEDURES FOH STAR/E/ LANDMARK SIGHTINGS - OF 3 MARKS EACH (ADDENDUM 4)		IGN NAV STARS INOPUS) IGEL) IGN CHECK NAV ST	ì
	RECOMMENDED NAV STARS NO. 15(SIRIUS) NO. 15(SIRIUS) NO. 16(PROCYON) RECOMMENDED LANDMARK		PROCEDURES FOR GOC ALIGN TO IMU (Addendum 3)	
	NO. 5(ALTERNATES=7,10,20,32,40.41,56,69,70,101,115,127,128)	(17+30) PROCEDURES HORIZON SI	RES FOR STAR/EARTH	
(03+60)	PROCEDURES DETERMINAT (ADDEND	TO JUNE OF THE PROPERTY OF THE	(P23) (ADDENDUM 4) (ACCOMMENDED NAV STARS NO. 22(REGULUS) (FAR HOR)	ē
(09+85)	PROCEDURES FOR ESTABLISHING P÷C (ADDENDUM 13)	ZZ ZZ	NO. 22(REGULUS) (FAR HOR) NO. 16(PROCYON) (FAR HOR) NO. 16(PROCYON) (FAR HOR) NO. 16(PROCYON) (FAR HOR)	
(12+00)	LMP RECEIVE BLOCK DATA PAD FROM GND	(14+20) PROCEDUR DETERM	PROCEDURES FOR LOI POSITION DETERMINATION	

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NO. 16 (PROCYON) (FAR HOR)	NO. 24 (SPICA) (NEAR HOR) (24+50)	PROCECURES FOR LOI POSITION DETERMINATION (ADDFNOUM 5) (P21)	(27+00) PROCEDURES FOR GND UPLINK OF CSM STATE VECTOR AND P30 TARGET LOAD (ADDFNHUM 8)	FOLLOWING GND UPLINK	LEB WHEN COMP LT-OUT(NO INTEG) KEY VOSE	F 16 54 (K.RDOT.THETA) COPY R AND RDOT	KEY VAGE	90		KEY V37EOOE	TRANSFER CSP	TRUM LM SLOTS TO CSM SLOTS)
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(P2)	ING PTC	) A C	ina PT.	3	, PS2)	α · · · · · · · · · · · · · · · · · · ·	•	ALIGN TO I.JU	Į.	3 SETS	(FCA)	S2
(ADDENDUM S)	PROCEDURES FOR ESTABLISHING (ADDENDUM 13)	LMP RECEIVE BLOCK NATA PAD FROM GND	PROCEDURES FOR TERMINATING (ADDENDUM 14)	DOOCEDINGS FOR TWI DAY 164	TO REFSMMAT  (ADDENDUM 2)	BACKUP ALIGN NAV STARS NO• 14 (CANOPUS) NO• 12 (RIGEL) BACKUP ALIGN CHECK NAV S	NO. 15(SIRIUS)	PROCEDURES FOR GNC ALIGN (ADDENNUM 3)	PROCEDURES FOR STABLEARTH	HORIZON SIGHTINGS + 3 :	(ADDENDUM 4)	RECOMMENDED NAV STARS
	(18+25)	(25+10)	(25+55)	(56+00)				(26+15)	(56+20)			

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PROCEDURES FOR STAR/EARTH HORIZON SIGHTINGS - 4 SETS OF 3 MARKS EACH (ADDENDUM 4)	RECOMMENDED NAV STARS NO. 16 (PROCYON) (FAR HOR) NO. 22 (REGULUS) (FAR HOR) NO. 21 (ALPHARD) (FAR HOR) (NO. 21 (ALPHARD) HAS MAG 2.2 AND MAY NOT BE VISIBLE.IF NOT SELECT NO. 22 (REGULUS))	R LOI POSITION	PROCEDURES FOR ESTABLISHING RIC (ADDENDUM 13)	PROCEDURES FOR TERMINATING PTC (ADDENDUM 14)	PROCEDURES FOR IMU REALIGN	TO REFSHMAT (ADDENDUM 2)	BACKUP ALIGN STARS NO. 14 (CANOPUS) NO. 12 (RIGEL)
		(59+00)	(50+05)	(33+55)	(34+00)		
PROCEDURES FOR EXT NV TARRETING (ADDENDUM 9)	######################################	TRIM DVS TO ZERO	PROCEDURES FOR SM RCS PROPULS-ON MONITOR CHECK (ADDENDUM 10)	PROCEDURES FOW RCS THRUST SET: P (ADDENDUM 12)	SECOND TRANSLUNAR MIDCOURSE	PROCEDURES FOR SM RCS AND SPS	(ADDENDUM 10)
(57+30)			(27+35)	(27+40)	(28+00)	(28+05)	(28+20)

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PROCEDURES FOR TERMINATING PTC (ADDENDUM 14)	PROCEDURES FOR IMU REALIGN TO REFSHMAT (ADDENDUM 2)	BACKUP ALION STARS NO. 14 (CANOPUS) NO. 12 (RIGEL) BACKUP ALION CHECK NAV STAR NO. 15 (STRTUS)	PROCE	PROCEDURES FOR STAR/LUNAR HORIZON SIGHTINGS 4 5 SETS OF 3 MARKS EACH (ADDENDUM 4)	RECOMMENDED NAV STARS NO. 33 (ANTARES) (NEAR HOR) NO. 33 (ANTARES) (NEAR HOR)	NO. 37 (NUNKI) (NEAR HOR) NO. 45 (FOMALHAUT) (FAR HOR) NO. 42 (PEACOCK) (FAP HOR)	PROCEDURES FOR LOI POSITION  OETERMINATION (ADDENNIM E)	
BACKUP ALIGN CHECK NAV STAR NO. 15(SIRIUS)	PROCEDURES FOR GOC ALIGN TO 1U (ADDENDUM 3)	PROCEDURES FOR STAR/EARTH HORIZON SIGHTINGS - 3 SETS OF 3 MARKS EACH (ADDENDUM 4)	RECOMMENDED NAV STARS NO. 16(PROCYON) NO. 22(REGULUS) NO. 26(SPICA)	PROCEDURES FOR LOI POSITION DETERMINATION (ADDENDUM 5) (ADDENDUM 5)	LMP RECEIVE BLOCK NATA PAN FROM GND	PROCEDURES FOH ESTABLISHING P+C (ADDENDUM 13)	LMP RECEIVE BLOCK DATA PAD FROM GND	
	(34+15) PR	(c c c c c c c c c c c c c c c c c c c		•	(34+55)	(35+00)	(00000)	(44+35)

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SFPS REQUIRES SCHEDULING THE MIDCOLRSE AT THE EARLIEST POSSIBLE TIME, HENCE THE SPS SHOULD NOT RE REQUIRED FOR THIS MIDCOURSE CORRECTION)	TRIM DVS TO ZERO	PROCEDURES FUR SM RCS PROPULSION MONITOR CHECK (ADDENDUM 10)	PROCEDURES FOR RCS THRUST SETUP (ADDENDUM 12)	THIRD TRANSLUNAR MIDCOURSE	PROCEDURES FOR SM RCS AND SPS PROPULSION MONITOR CMECKS (ADDENDUM 10)	<b>B.</b>	RECOMMENTED NAV STARS NO. 16 (PROCYON) (FAR HOR) NO. 22 (REGULUS) (FAR HOR) NO. 26 (SPICA) (NEAR HOR)
		(46+35)	(04.44)	(47+00)	(47+05)	(47+25)	
(46+00) PROCEDURES FOR GND UPLINK OF SM STATE VECTOR AND P30 TARGET 1 0AD (ADDENDUM B) FOLLOWING GND UPLINK	POO LEH WHEN COMP LT-OUT (NO INTEGNET NOTE NOTE NOTE NOTE NOTE NOTE NOTE N	F 16 54 (R. NOOT THETA) COPY R AND RDOY PRO	KEY VUGE F 04 05 (TRACK AXIS OPTION) LOAD 00002 IN R2	F 06 18 (COMPUTED R.P.Y) 20PY R.P.Y KEY V34E	POO KEY V37EOOE KEY V47E (TRANSFER CSM SV FEOM 12 SLOTS 20 SE 21 SE	08 Ext 0V 1/	PROPULSION SYSTEM SELECTION USE RCS FOR DV GREATER THAN 1FPS AND LESS THAN OR EQUAL TO SFPS (A DV GREATER THAN

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OF 3 MARKS EACH (ADDENDUM 4) (P23)	NO. 37 (NUNKI) (NEAR HOR)	NO. 33 (ANTARES) (NEAR HOR) NO. 33 (ANTARES) (NEAR HOR) NO. 33 (ANTARES) (NEAR HOR)	PROCEDURES FOR LOI POSITION DETERMINATION (ADDENNUM 5)	ESTABLISHING P	PROCEDURES FOR TERMINATING PTC (ADDEWOUM 14)	PROCEDURES FOR GND UPLINK OF CSW STÂTE VECTOR.P30 TARGET LOAD.AND PREFERREC LUNAR ORBIT REFSMMAT (ADDENDUM 8)	LMP RECEIVE GND BLOCK DATA PAD	PROCEDURES FOR IMU ALIGN TO UPLINKED PREFERRED REFSMMAT (ADDENDUM 2)
			(93+69)	(52+5)	(54+25)	(50+30)	(00+09)	
PROCEDURES FOR LOI POSITION		PROCEDURES FOR ESTABLISHING P+C (ADDENJUM 13)	LMP RECEIVE BLOCK NATA PAD FROM GND	PROCEDURES FOR TERMINATING PTS (ADDENDUM 14)	PROCEDURES FOR IMU REALIGN To refsmmat (addendum 2)	BACKUP ALIGN NAV STÄRS NO. 14 (CANOFUS) NO. 12 (RIGEL) RACKUP ALIGN CHECK NAV STAR NO. 15 (SIRIUS)	PROCEDURES FOR GDC ALIGN TO INU (ADDENDUM 3)	PROCEDURES FOR STAR/LUNAR Horizon Sightings - 9 Sets
(47+55)		(48+00)	(51+00)	(\$2+€\$)	(52+10)		(\$2+28)	(52+30)

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FOURTH TRANSLUNAR MINCOURSE	PROCEDURES FOR SM PCS AND SPS PROPULSION MONITOR CHECKS (ADDENDUM 10)	PROCEDURES FOR LOII POSITION		PROCEDURES FOR ESTABLISHING PTC (ADDFNDUM 13)	PROCEDURES FOR TERMINATING PTC (ADDENDUM 14)	PROCEDURES FOR IMU REALIGN	(ADDENOUM 2) BACKUP ALIGN NAV STARS	NO. 12(RIGEL) BACKUP ALIGN CHECK NAV STAR NO. 15(SIRIUS)	PROCEDURES FOR GDC ALION TO IMU (ADDENDUM 3)	LMP RECEIVE LOII PAD FROM 6ND
	(60+19)	(6]+10)	(6]+15)		(65+55)	( <sub>0</sub> 0+9 <u>-</u> 9)			(66+15)	(66+20) Poo
BAC':UP ALIGN NAV STARS NO. 14 (CANOPUS)	NO. 12(RIGEL) PACKUP ALIGN CHECK NAV STARS NO. 15(SIRIUS)	PROCEDURES FOR GOC ALIGN TO 1"U (ADDENDUM 3)	PROCEDURES FOR EXT DV TARGETING (ADDENDUM 9)	SOUTH A SERVICE OF THE PROPERTY OF THE PROPERT	S FOR DV TO SFPS SFPS WILL	FER	TRIM DVS TO ZERO	PROCEDURES FOR SM ACS PROPULSTON MONITOR CHECK (ADDENDUM 10)	PROCEDURES FOR RCS THRUST SETTING	ì
		(60+25)	(60+30)					(60+35)	(60+40)	(61+00)

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OPT YERO-OFF  OPT MODE-CHC KEY V41N01E  KEY V41N01E  F 21 92 (8LANK, 8LANK)  LOAD PAD SHAFT ANGLE(+XXX, XX)  IN R1 AND PAD TRUNNION ANGLE  (+XX, XXX) IN R2  (+XX, XXX) IN RX  (+XX, XXX) IN RXINFS  OPT YERO-ZERO  RETICLE BRT TW-MIN RRINFS	(66+30) PROCEDURES FOR ESTABLISHING PTC (ADDENDUM 13)	(67+30) PROCEDURES FOR TERMINATING PTC (ADDENDUM 14)	(\$7+35) PROCEDURES FOR GDC DRIFT CMECK (ADDFNDUM 7)	PROCEDURES FOR IMU REALIGN TO REFSMMAT FOR DRIFT CHECK (ADDENDUM 2)	HACKUP ALIGN NAV STARS NO. 14 (CANOPUS) NO. 12 (RIGEL)
č	LEB KEY V49E F 06 22 (FINAL GMBL ANGLES) LOAU BURN ATT STAR CHECK GMBL ANGLES KEY V62E	~ ~	CDP MONITOR ATT MANEUVER FOR GWBL LOCK F 50 18 (COMMANDEU R.P.Y) VERIFY R.P.Y AGREE WITH PAU R.P.Y WITHIN 5 DFG	NULL FDAI NEEDLES WITH RHALER PROGN PPROGNES	OPT ZERO-ZERO(15SEC) OPT TELTRUN-SLAVE TO SXT RETICLE BAT TW-ADJUST

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(68+20) CDR RECEIVE GO/NO-GO FOR LOII PROCEDURES FOH GND LIPLINK OF ASTATE VECTOR AND P30 TARGET 10 (ADDENDUM 8) LMP INITIATE PRE-LOI SYSTEMS AH (ABTERMINATION FORTEMS AND PSOCEDURES FOR EXT DV TARGETING (ADDENDUM 9)  ***********************************
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VERIFICATION (ADDENDUM 18)	LMP UP TLM-CMB RESET UP TLM-NORMAL	MSFN LOS (70+55)	CDR MAN ATT(P)-ACC CMD MANELVER SC TO 315 DEG PITCH	AND 180 DEG ROLL W.R.T. LOCAL HORIZONTAL MAN ATT(P)-MIN IMP INITIATE ORBITAL PITCH RATE	SUNIJP (7)+00)	X C	FAMILIARIZATION IDENTIFY IDENTIFICATION POINTS AND CONTROL POINTS AT RESPECTIVE TIMES OF CLOSEST	APPROACH	*** REGIN REV 2 *** (71+21)	(71+30) **PREPARE FOR TV PICTURES**	
MANEUVER SC IN BOLL 190 DEG	MAN ATT(3) -RAIF CMU LMP SELECT S BO ANT OMNI-A OR B	OVBU-ON Sh Ranging-Off	MSFN AOS (69+30)	PROCEDURES FOR HI GAIN ACQUISTION OF MSFN (ADDENDUM 19)	LMP RECEIVE BLOCK DATA FOR TEI FROM MSFN	SUNDOWN (70+14)	PROCEDURES FOR IMU REALIGN TO REFSMMAT (ADDENDUM 2)	RACKUP ALIGN NAV STARS NO. 14 (CANOPUS)	RACKUP ALIGN CHECK NAV STAR NO. 15(SIRIUS)	(70+30) PROCEDURES FOR GDC ALIGN TO IUU (ADDENDUM 3)	PROCEDURES FOR ORDEAL

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CDR MAN ATT(Y) -ACCEL CMD  YAW SC RIGHT TO 45 DEA YA  315 DEG PITCH AND 180 DEA  ROLL WAST. LOCAL HORIZOTAI  MAN ATT(Y) -RATE CMD  MAINTAIN ORBITAL PITCH RATE  FOR TV PICTURES  COR TV PICTURES  CON MAINTAIN ORBITAL PITCH RATE  (72.22)  (72.42)  (72.430)  (72.45)  COR MAN ATT(Y) -ACCEL CMD  YAW SC LEFT TO 0 DEG YAW  315 DEG PITCH AND 180 DEA  ROLL WAST. LOCAL HORIZOTAL  MAINTAIN ORBITAL PITCH RATE  (72.45)  CMP PSEUDO LANDING SITE  FAMILLARIZATION  IDENTIFY IDENTIFICATION PAINTS	AND PSEUDO LANDING SITE AT RESPECTIVE TIMES OF CLOSEST APPROACH COR STOP PITCH RATE BY SWITCHING MAN ATT(3)-RATE CMD	LMP RECEIVE BLOCK DATA FOR	PROCEDURES FOR IMU REALIGN TO REFSHMAT (ADDENDUM 2)	BACKUP ALIGN NAV STARS NO. 14 (CANOPUS) NO. 12 (RIGEL) BACKUP ALIGN CHECK NAV STAR NO. 15 (SIRIUS)	PROCEDURES FOR GOC ALIGN TO IMU (Addendum 3)	PROCEDURES FOR EXT DV TARGETING (ADDENDUM 9) ************************************
	MAN ATT(Y)-ACCEL CMD YAW SC RIGHT TO 45 DEG YA 315 DEG PITCH AND 180 DEG ROLL W.R.T. LOCAL HORIZOTA; MAN ATT(Y)-RATE CMD MAINTAIN ORBITAL PITCH RATE FOR TV PICTURES SELECT S BD ANT OMNI-A OR B	ING OFF	HI GAIN ACQUISTION 19)	AND P30 TARGET ! OAD 8) -ACCEL CMD T TO 0 DEG YAW	315 DEG PIICH AND 180 DEG ROLL W-R-T. LOCAL HORIZOMTAL Man att(Y)-Rate CMD Maintain orbital Pitch Rafe	PSEUDO LANDING SITE FAMILIARIZATION IDENTIFY IDENTIFICATION PAINTS

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ā	PROPULSION SYSTEM SELECTION	(ADDENDUM 10)
	CSE SDS/DO NOT TRIM DV CARACACACACACACACACACACACACACACACACACAC	PROCEDURES FOR GDC ALIGN TO IMU (ADDENDUM 3)
(CC+S7)	PROCEDURES FOR SM RCS PRODULS+ON MONITOR CHECK (ADDENDUM 10)	(73+40) PROCEDURES FOR ORDEAL SETUP (ADDENDUM 18)
Ţ	LMP UP TLM-CMU RESET UP TLM-NORMAL	(73+45) CDR ATT DB=MAX RATE=LOW
MSFN LOS (73+03)		BMAG MODE(3)-ATT 1/BATE 2 SC CONT-SCS MAN ATT(8.8)-ACCFL CMD
α <b>0.</b>	PROCEDURES FOR SPS THRUST SETHP (ADDENDUM 11)	MAN ATT (Y) - RATE CMD MANEUVER SC TO 307 DEG PITCH AND 140 DEG ROLL W.R.T.
SUNUP (73+09)		E C F
(73+31) SE	SECOND LUNAR ORBIT INJECTION	INTIATE ORBITAL PITCH
3	LMP TLM INPUTS PCM-LBR TAPE RCDR-FWD	LMP SELECT S BD ANT UMNITA ON BOVBULON SB RANGING-OFF
		MSFN AOS (73+47)
*** BEGIN B (73+31)	### € ∧⊒a	PROCEDURES FOR HI GAIN ACQUISITION OF MSFN (ADDENDUM 19)
(73+75) pq	PROCEDURES FOR SM RCS AND SPS PROPULSION MONITOR CHECKS	(74+15) CDR WHEN PSEUDO LANDING SITE IS

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BEGIN PHOTOGRAPHIC PREPARATION FOR VERTICAL STEREO NAVIGATION PHOTOGRAPHYY UP TLM-CMD RESET UP TLM-NORMAL	MAN ATT(P)-ACCEL CMD PITCH SC UP TO 270 DEG P AND 180 DEG ROLL W.R.T. LOCAL HORIZONTAL MAN ATT(P)-MIN IMP	INITIATE ORBITAL PITCI	KEY VOGNGS WHEN X-AXIS CROSSES TERMINATOR KEY ENTER/START CAMERA FOR VERTICAL STEREO NAVIGATION	O & X	(R,P.Y) RECORD PRESENT GMBL ANGLES MONITOR PHOTOGRAPHY
74+85) LMP LOS	C0R	SUNUP (75+09)	(75+13) (75+15) (OR+LMP	00 00 00 00 00 00 00	06 20 LAP
ACGUIDED IN COAS-INCREASE THE PITCH RATE TO KEEP X-AXIS CAMERAS POINTED AT IT TERMINATE PITCH RATE WHEN LANDMARK IS NO LONGER VIABLE COR MAN ATT(P)-ACCEL CMD PITCH SC DOWN TO 0 DEG PITCH AND 190 DEG ROLL W.R.T. LOCAL HORIZONTAL	LMP RECEIVE BLOCK NATA FOR Tel From MSFN	EALIGN (PS2)	BACKUP ALIGN NAV STARS NO 14 (CANOPUS) NO 12 (RIGEL) BACKUP ALIGN CHECK NAV STAR NO 15 (SIRIUS)	(74+45) PROCEDURES FOR GDC ALIGN TO INU (ADDENDUM 3)	(74+50) PROCEDURES FOR ORDEAL VERIFICATION (ADDENDUM 18)

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CDR MAN ATT(P).A.CEL CMN PITCH UP TO 5 DEG PITCH AND 0 DEG ROLL W.R.T. LOCAL HORIZONTAL MAN ATT(P).MIN IMP INITIATE ORBITAL PITCH RATE	##BEGIN LANDING SITE LIGHTING EVALUATION** LEB KEY V37E22E  F G6 45 (BLANK,BLANK,MAX MGA) LEB PRO F 05 70 (LANDMARK CODE) LOAD 10001 IN R2 LEB OPT 7ERO-OFF	POSITION TRUNNION TO 10 DEG LEB IDENTIFY IDENTIFICATION POINTS AND PSEUDO LANDING SITE AT THETA RESPECTIVE ACQUISITION TIMES EVALUATE THE LIGHTING BETWEEN TERMINATOR LEB UPON ACQUISITION OF THE LEN PRO OPT MODE-CMC LEN PRO O6 92 (SHAFT, TRUN, BLANK) TION OF THE PSEUDO LANDING
CDR MONITOR ORBITAL PITCH RATE  ** BEGIN REV & ***  (75+30)	(75+40)  COR MAN ATT(R)-ACCEL CMD  AT THE SUBSOLAP POINT— ROLL SC 180 DEG TO 0 DEG RO! L  AND 270 UEG PITCH W.R.T. LOCAL HORIZONTAL  MAN ATT(R)-RATE CMD  REINITIATE ORBITAL PITCH ATF  LMP SELECT S BU ANT OMNI—C OR U  DVRU—ON  SR RANGING-OFF	(75+55)  (75+55)  (75+55)  CDR KEY VOGN65  CDR,LMP KEY ENTER/STOP CAMERA  06 65 (GET)  LMP RECORD GET  CDR KEY N20E  06 20 (R*P*Y)  LMP RECORD PRESENT GMBL ANGLEC  CDR STOP PITCH RATE  (76+00) - (TCA OF 70 DEG LONGITH) DE WILL  RE UPDATEN REALTIME)

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LMP RECEIVE RLOCK DATA FOR TEI FROM MSFN	('A+3U) PROCEDURES FOR IMU REALIGN TO REFSWMAT (ADDENDUM 2) (P52)	BACKUP ALIGN NAV STARS NO 14 (CANDPUS) NO 12 (RIGEL) BACKUP ALIGN CHECK NAV STAR NO 15 (STRIUS)	(76+45) Procedures for GDC Align to Imu (Addendum 3)	(76+50) PROCEDURES FOR ORDEAL VERIFICATION (ADDENDUM 18)	LMP UP TLM-CMD RESET UP TLM-NORMAL	MSFN LOS (76+58)	CDR MAN ATT(R,P)-ACCEL CMD MANEUVER SC TO S DEG PITCH AND 0 DEG ROLL W.R.T. LOCAL HORIZONTAL MAN ATT(R)-RATE CMD	MAN ATT(P)-MIN IMP INITIATE ORBITAL PITCH RATE FOR LANDMARK SIGHTINGS	26
HAFT AXIS OF LANDING SITE AD . ONG.D	VISABLE- OPT MODE-MAN EVALUATE EARTHSHINE TRACK+NG FEASABILITY	LEB POSÍTION TRUNNTON TO 5 DEA OPT ZERO-ZERO OPT MODE-MAN LEB KEY V37E00E		EG PITCH AND 180 DEG - W.R.T. LOCAL HORIZONTAL ATT(R)-RATE CMD		PROCEDURES FOR HI GAIN ACQUISTITON OF MSFN (ADDENDUM 19)	PROCEDURES FOR GND UPLINK OF FSM State Vector (addendum B)	CMP KEY V47E	

(76+20)

SUNDOWN

(12+91)

(76+25)

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9 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	SUNDEN
2 2	PROCEDURES FOR HI GAIN ACQUISITION OF MSFN (ADDFNDUM 19)
PROCEDURES FOR TRACKING THE SECOND CONTROL POINT LANDMARK (NO AUTO OPTICS) (ADDENDUM 16)	(78+25) PROCEDURES FOR GND UPLINK OF CSM STATE VFCTOR (ADDENDUM 8)
** BFGIN REV 5 *** (77+29)	CMP KEY V47E LMP RECEIVE BLOCK DATA FOR
LMP SELECT S BD ANT OMNITO OR U DVBULON SR RANGING-OFF	
1STN AOS (77+44)	HACKUP ALIGN NAV STARS
(78+05) PROCEDURES FOR TRACKING THE PSEUDO LANDING SITE (ADDENDUM 15)	NO 14(CANOPUS) NO 12(RIGEL) HACKUP ALIGN CHECK NAV STAR NO 15(SIRIUS)
CMD TO ABOUT	(7A+45) PROCEDURES FOR GDC ALIGN TO IMU (ADDFNDUM 3)
DFG PITCH W-R-I. LO	(7a+50) PROCEDURES FOR ORDEAL VERIFICATION (ADDFNDUM 18)
(78+20)	LMP UP TLM-CMD RESET

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(8n+00) PROCEDURES FOR TRACKING THE PSEUDO LANDING SITE (ADDFNDUM 15)	(Bn+15) CDR MAN ATT(R)-ACCEL CMN ROLL SC 180 DEG TO ABOUT 300 DEG PITCH AND 180 DEG ROLL W.R.T. LOCAL MORIZONTAL MAN ATT(R.P)-RATE CMD	PROCEDURFS FOR HI GAIN ACQUISITION OF MSFN (ADDENDUM 19) (80+25) PROCEDURES FOR GND UPLINK OF 25M STATE VFCTOR (ADDENDUM 8)	CMP KEY V47E LMP RECEIVE BLOCK DATA FOR TEI FROM MSFN	(8n+30) PROCEDURES FOR IMU REALIGN TO REFSMMAT (ADDENDUM 12)	BACKUP ALIGN NAV STARS
(1) TLM_NORMAL (8) (8) (78+54)	CDR MAN ATT(R.P)-ACCEL CMD MANEUVER SC TO 5 DEG PITCL AND 0 DEG ROLL W.R.T. LOC.L HORIZONTAL MAN ATT(R)-RATF CMD MAN ATT(R)-MIN IMP INTITATE ORBITAL PITCH RATE FOR LANDWARK SIGHTINGS	(79+25) (79+25) PROCEDURES FOR TRACKING THE SECOND CONTROL POINT LANDMARK (WITH AUTO OPTICS) (AITH AUTO OPTICS) (ADDENDUM 17)	*** BFGIN REV 6 *** (79+28)	LWP SELECT S BD ANT OWNI-C OR D DVBU-ON SB RANGING-OFF	179+2)

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(81-25) PROCEDURES FOR TRACKING THE SECOND CONTROL POINT LANDMARK (WITH AUTO OPTICS) (ADDENDUM 17) (P22)	+++ REGIN REV 7 +++ (81+27) LMP SELECT S BD ANT OMNI-C OR D	MSFN AOS (81+40)	THIRD CONTROL POINT LANDMARK (NO AUTO OPTICS) (ADDFNDUM 16) (B2.00)	PROCEDURES FOR TRACKING THE PSEUDO LANDING SITE (ADDENDUM 15)	CDR MAN ATT(R)-ACCEL CMD ROLL SK 180 DEG TO ABOUT 300 DEG PITCH AND 180 DEG ROLL W.R.T. LOCAL HORIZONTAL MAN ATT(R.P)-RATE CMD
NO 12(RIGEL) BACKUP ALIGN CHECK NAV STAM NO 15(SIRIUS) (80+45) PROCEDURES FOH GDC ALIGN TO IU (ADDENDUM 3)	(80+50) PROCEDURES FOR ORDEAL VERIFICATION (ADDENDUM 18) LMP UP TLM-CMU RESET	MSFN LOS 180+54) COR MAN ATT(R+P)-ACCEL CMD	MANEUVER SC TO 5 DEG DITCE AND 0 DEG ROLL W.R.T. LOCAL HORIZONTAL MAN ATT(R) RATE CMD MAN ATT(P) MIN IMP	SUMUP (81+04)	(81+10) PROCEDURES FOR TRACKING THE FIRST CONTROL POINT LANDWARK (NO ALITO OPTICS) (ADDENDUM 16)

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TLM-CMD RESET	TIMONOPHAL
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LMP UP TLM-CMD RESET UP TLM-NOPMAL	MSFN LOS (82+52) COR MAN ATT(R.P).ACCEL CMD MANEUVFR SC TO 5 DEC PITCH	AND O DEG ROLL W.R.T. LOCAL HORIZONTAL MAN ATT(R)-RATE CMD MAN ATT(P)-MIN IMP INITIATE ORBITAL PITCH RATE	FOR LANDMARK SIGHTINGS SUNUP (83+03)	PROCEDURES FOR TRACKING THE FIRST CONTROL POINT LANDMARK (WITH AUTO OPTICS) (ADDENDUM 17)	(81+20) PROCEDURES FOR TRACKING THE SECOND CONTROL PCINT LANDMARK (WITH AUTO OPTICS) (ADDENDUM 17) (P22)	*** DEGIN REV B ***	(83+26)
SUNDOWN (82+17)	PROCEDURES FOR HI GAIN ACOUISTION OF MSFN (ADDENDUM 19)	PROCEDURES FOR GND UPLINK OF ASM STATF VECTOR (ADUENDUM 8)		(82+30) PROCEDIFIES FOR IMU REALIGN TO REFSMMAT (ADDENDUM 12) (P5.5)	HACKUP ALIGN NAV STARS NO 14 (CANOPUS) NO 12 (RIGEL) RACKUP ALIGN CHECK NAV STAR NO 15 (SIRIUS)	(82+45) PROCEDURES FOR GDC ALIGN TO INU (ADDENDUM 3)	(82+50) PROCEDURES FOR ORDEAL VERIFICATION (ADDENDUM 18)

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LMP RECEIVE RLOCK DATA FOR TEI FROM MSFN	(84+25) PROCEDURES FOR IMU REALIGN TO REFSMMAT (ADDENDUM 12) (P52)	HACKUP ALIGN NAV STARS No 14 (CANOPUS)	3	PROCEDURES FOR GDC ALIGN TO IMU (ADDENDUM 3)	PROCEDURES FOR ORDEAL VERIFICATION (ADDENDUM 18)	LMP REGIN PHOTOGRAPHIC PREPARATION For darkside and solar corona Photograpky	CDR MAN ATT(P)-ACCEL CMD PITCH SC UP TO 0 DEG PITCH AND 180 DEG ROLL W.R.T. LOCAL HORIZONTAL	MAN ATT(P)=MIN TMP INITIATE ORBITAL PITCH RATE	(84+45) Lmp Perform Darkside and Solar Corona Photography
DVBU-ON SA RANGING-OFF	MSFN AOS (83+39) FROCEDURES FOH TRACKING THE THIRD CONTROL POINT LANDMARK	(WITH AUTO OPTICS) (ADDENDUM 17)	(84+00) PROCEDURES FOR TRACKING THE PSEUDO LANDING SITE (ADDENDUM 15)	(94+10) CDR MAN ATT(R)_ACCFL CMD	ROLL SC 180 DEG TO ABOUT 300 DEG PITCH AND 180 DEA ROLL WORT - LOCAL HORIZONTAL	A + + + + + + + + + + + + + + + + + + +	PROCEDURES FOR HI GAIN ACOUISTION OF MSFN (ADDENDUM 19)	(84+20) PROCEDURES FOR GND UPLINK OF ASM	CMP KEY V47E

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	444 HFGIN REV V 444 (81425)	ARATION (85+35) COR PITCH	¥ .	NOW NACE	E O	ON MC	E CONTRACTOR
UP TLM_CMD RESETUP TLM_NORMAL		BEGIN PHOTOGRAPHIC PREP FOR CONVERGENT STERED NAVIGATION PHOTOGRAPHY MAN ATT(P) "ACCFL CMD PITCH SC UP TO 290 DEG AND 180 DEG ROLL W.R.T LOCAL HOHIZONTAL MAN ATT(P) "MIN IMP		KEY VOGN65	KIEN XAAXIS KEY ENTER/S CONVERGENT PHOTOGRAPHY	(GET) RECORD GET KEY N20E (R.P.Y)	PHO
Z Z	48FN LOS (84+51)	(85+00) LMP CDR	_	(85+05) LMP	CDR.LMP	06 C 65 06 06 06	0 Q

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CDR MONITOR ATT MANEUVER	F 50 18 (COMMANDED R.P.Y)	PAD B.P. VITILS SOFG	NULL FDAT NEEDLES WITH RHC LEB PRO	G/N PWR.OPTICS=ON	OPT ZERO-ZERO(15SEC) OPT TELTBINS AVE TO SXT	RETICE BRY TWANDUST	OPT MODELCING	KEY VAINGIE	F 21 92 (BLANK, BLANK, BLANK)	CXX-XXX+) MJSXX - MILD OXX OXOJ P EVX - XXIIVI OXA OXX OXX OXX OXX OXX OXX OXX OXX OXX	(+XX*XX) IN BO	41 88 (SHAFT, TRUNNION, BLANK)	MONITOR OPT DRIVE TO PAD	SPITE	OPT NODELEAN	DRIVE TRUN LESS THAN 5 DEG		SOURCE STREET THE THINK	(84+05)	PROCEDURES FOR GDC DRIFT CHECK	(ADDENDUM 7)		(84.10) COR RECEIVE GO.NO.GO FOR TEI	
PROCEDURES FOR HI GAIN ACQUISTION			PROCEDURES FOR GND UPLINK OF ASM STATE VECTOR AND P30 TARGET LOAD		LMP RECEIVE BLOCK DATA FOR TFI FROM MSFN	24 + 4 + 4 + 4 + 4 + 4 + 4 + 4 + 4 + 4 +	PROCEDURES FOR TEI BURN ATT	CHECK EXCEPT ROLL	COB BOAT SCALES	FDAI SELECT-1/	MAN ATT(3) -RATE CMD	10-373/	ATT DOSEMAX	APATE LOS	CHOPELAUTO	AMAG MODE (3) -RATE2	RHC-ARMED	LEB KEY V49E	(FINAL GMBL A	LOAD BURN A	CK GMBL A	KEY VOZE	F 50 18 (COMMANDED R.P.Y)	PRO

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SO	PROCEDURES FOR SPS THRUST SETUP (ADDENDUM 11)		TRANSEARTH INJECTION	PROCEDURES FOR SM RCS AND SPS PROPULSION MONITOR CHECKS	CDR MAN ATT (P) -ACC CMD RHC-ARMED	MANEUVER SC IN PITCH(UP) TO MONITOR MOON IN CMD WINDOW IAN ATT(3) - RATE CMD	PROCEDURES FOR HIGH GAIN ANT ACQUISITION	PROCEDURES FOR TO REFSHMAT	
MSFN LOS (89+48)	(88+45)	SUNIP (84+59)	(89+15)	(89+20)	000		MSFN A0S (8c+20)	(00+06)	
LMP INITIATE PRE-TEI SYSTEMS CHECK		PROCEDURES FOR IMU REALIGN TO REFSMMAT FOR DRIFT CHECK (ADDENDUM 2)	BACKUP ALIGN NAV STARS NO. 14 (CANOPUS)	BACKUP ALIGN CHECK NAV ST.R NO. 15(SIRIUS)	PROCEDURES FOR GDC ALIGN TO INU (ADDENDUM 3)	PROCEDURES FOR EXT DV TARGETING (ADDENDUM 9)	PRODUCION SYSTEM SELECTION USE SPS AND TRIM DVS TO SEPS ***********************************	PROCEDURES FOR SM RCS PRODULSTON MONITOR CHECK (ADDENDUM 10)	
	SUNDOWN (88+13)				(88+30)	(88+35)		(88+4)	

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PROCEDURES FOR TERMINATING F (ADDFNDUM 14)	PROCEDURES FOR IMU REALIGN	(ADDENDUM 2) BACKUP ALIGN NAV STARS	BACKUP ALTON CHECK NAV (	PROCEDURES FOR GDC ALIGN TO (ADDFNDUM 3)	PROCEDURES FOR STAR/LUNAR HORIZON SIGHTINGS - 3 SETS OF SMARKS EACH	RECOMMENDED NAV STARS NO. 1 (ALPHERATZ) (NEAR NO. 11 (ALDEBARAN) (FAR NO. 2 (DIPHDA) (NEAR	PROCEDURES FOR STAR/EARTH HORIZON SIGHTINGS - 6 SETS OF 3 MARKS EACH (ADDENOUM 4)	RECOMMENDED NAV STARS
	(100+10)		•	(100+25)	(100+30)		(101+20)	
	STAR	n·1	•	(E) (A)			(131)	
NO. 14 (CANOPUS)	NO. 12 (RIGEL) BACKUP ALIGN CHECK NAV S NO. 15 (SIRIUS)	PROCEDURES FOR GDC ALIGN TO (ADDENDUM 3)	PROCEDURES FOR STAR/LUNAR HORIZON SIGHTINGS - 8 SETS OF 3 MARKS EACH	ži G G	1 (ALDERATZ) (NEL 10 (MIRFAK) (FAMINETAK) (FAMINETAK) (FAMINETAK) (FAMINETAK) (FAMINETAK) (FAMINETAK) (FAMINETAK) (FAMINETAK)	PROCEDURES FOR EST	PROCEDURES FOR EARTH TARGETI	
		(90+15)	(90+30)			(91+50)	(91+55)	(100+02)

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(102+20) (103+20) (103+20)	KEY V47E (TRANSFER CSM SV From Lm Slots to CSM Slots)	PROCEDURES FOR IMU ALIGN TO	UPLINKED PREFERRED REFSMMAT (PS2)	BACKUP ALIGN NAV STARS NO. 14 (CANOPUS)	BACKUP ALIGN CHECK NAV STAR NO. 15(SIRIUS)	PROCEDURES FOR GDC ALIGN TO IMU	•	PROCEDURES FOR EXT DV TARGETING		PROPULATION ANATEM AFFECTION	USE RCS FOR DV LESS THAN OR	EQUAL TO 12FPS	TIAN PLOTON OF GREATER		TRIM DVS TO ZERO			PROCEDURES FOR SM RCS PROPULSION	(ADDENDUM 10)
Z-20)  ROCEDURES FOR RETURN TO EARLY OF STATE VECTOR, PROCEDURES FOR RETURN TO EARLY OF STATE VECTOR, P30 TARGET LOAD PROCEDURES FOR GND UPLINK OF STATE VECTOR, P30 TARGET LOAD PREFERED ENTRY REFSMMAT (ADDENDUM 9)  FOLLOWING GND UPLINK OF STATE VECTOR, P30 TARGET LOAD PREFERED ENTRY REFSMMAT (ADDENDUM 9)  FOLLOWING GND UPLINK OF KEY V83E  KEY V83E  F 16 54 (R,RDOT,THETA)  COPY R AND RDOT  PRO KEY V83E  F 04 06 (TRACK AXIS OPTION)  LOAD 0,002 IN R2  PRO KEY V34E  KEY V34E		(103+10)				(103+25)	105,501,										(103+35)	,	
	. 22(REGULUS) (FAR ) . 22(REGULUS) (FAR ) . 26(SPICA) (NEAR	• 26(SPICA) (NEAR • 31(ARCTURUS)(NEAR	. 31 (ARCTURUS) (NEAR	ETC	(ADDENDUM 6)	PROCEDURES FOR GND STATE VECTOR, P30 T	PREFERRED ENTRY REFISHMAT	ANT IGH GNO SUPPLIED	POO LEB WHEN COMP LT-011 (NO INTEG.	XEY V83E	•	COPY R AND ROOT	KEY V89E	04 06 (TRACK AXIS	1040 00002 IN R2	06 18 (COMPUTED	COPY R.P.Y	KEY V3♠E	

FINAL MISSION C-PRIME TRANSLUNAR. LUNAR PARKING ORRIT. AND TRANSEARTH TIMELINE -12 NOV 1968

PROCEDURES FOR RETURN TO EARTH TARGETING (ADDENDUM 6)	PROCEDURES FOR IMU REALIGN To berammat	(ADDENDE 2) (ADDENDE 2) (ASZ)  BACKLIP ALIGN NAV STARS	NO. 14(CANOPUS) NO. 12(RIGEL) BACKUP ALIGN CHECK NAV STAR NO. 15(SIRIUS)	PROCEDURES FOR GDC ALIGN TO IMU (ADDENDUM 3)	PROCEDURES FOR STAR/EARTH HORIZON SIGHTINGS - 5 SETS OF 3 MARKS EACH	NAV STARS LUS) (FAR T	NO. 25 (SPICA) (NEAR YOR) NO. 31 (ARCTURUS) (NEAR HOR) NO. 31 (ARCTURUS) (NEAR HOR)	PROCEDURES FOR ESTABLISHING PTC (ADDENDUM 13)
(106+45)	URSF (107+40)	S AND SPS CHECKS	EARTH 5 Sets		(NEAR HOD) (NEAR HOD) (NEAR HOD)	UNAR 3 SETS ,P23)	~ .	
PROCEDURES FOR RCS THRU (ADDENDUM 12)	FIRST TRANSEARTH MIDCOURSF	PROCEDURES FOR SM RCS A PROPULSIÓN MONITOR CHE (ADUE JUM 10)		S	NO. 26 (SPICA) NO. 31 (ARCTURUS) (N NO. 31 (ARCTURUS) (N	PROCEDURES FOR STAR/LUNHORIZON SIGHTINGS - 3 OF 3 MARKS EACH (ADUENDUM 4)	RECOMMENDED NAV STARS NO. 1 (ALPHERATZ) (NEAR HAR NO. 11 (ALDERARAN) (FAR HOD)	
(103+40)	(104+05)		(105+15)		0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0			

FINAL MISSION C-PRIME TRANSLUNAR, LINAR PARKING ORRIT, AND TRANSEARTH TIMELINE -12 NOV 1968

DPOCEDURES FOR RETURN TO EARTH TARGETING (ADDENDUM 6)	D) PROCEDURES FOR GND UPLINK OF CSM STATE VECTOR AND P30 TARGET LOAD (ADDENDUM 8) (P30)	3 8 9	90 90 90 90	06 18	KEY V37E00E KEY V47E (THANSFER CSM SV FROM LM SLOTS TO CSM SLOTS)	PROCEDURES FOR EXT DV TARGETING (ADDENDUM 9)  ***********************************
(120+30)	(121+00)	000	a. U.	la.	000	(121+30)
H RETURN TO ING 6)	ERWINATING PT	FOR IMU REALIGN IT IUM 2)	BACKUP ALIGN NAV STARS NO. 14 (CANOPUS) NO. 12 (RIGEL) RACKUP ALIGN CHECK NAV STAR NO. 15 (SIRIUS)	R GDC ALIGN TO I U	H STAH/EARTH Tings - 3 Sets Ach 4)	STARS (FAR HOR (NEAR HO) (NEAR HO)
(108+55) PROCEDURES FOR R EARTH TARGETING (ADDENDUM 6)	(119+35) PROCEDURES FOH TERM; (ADDENDUM 14)	(119446) PROCEDURES FOR TO REFSMMAT (ADDENDUM	PACKUP ALIGN NAVNO. 14 (CANOPUS) NO. 12 (RIGEL) RACKUP ALIGN CHE	(119+55) PROCEDURES FOR GDC (ADDENDUM 3)	(120+00) PROCFDURES FOH STAH/EARTH HORIZON SIGHTINGS = 3 SE OF 3 MARKS EACH (ADDENDUM 4)	RECOMMENDED NAV NO. 22 (REGULUS) NO. 26 (SPICA) NO. 31 (AMCTURUS

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FINAL MISSION C.PRIME TRANSLUNAR. L'ARR PARKING ORRITIAND TRANSEARTH TIMELINE -12 NOY 1968

(123+35) PROCEDURES FOR STAR/EARTH HORIZON SIGHTINGS - 5 SETS OF 3 MARKS FACH	(ADDENDUM 4) RECOMMENDED NAV STARS NO. 22 (REGULUS) (FAR HOR)	22 (AFGULUS) (FAR 26 (SPICA) (NEAR 31 (ARCTURUS) (NFAR 31 (ARCTURUS) (NFAR	(124+25) PROCEDURES FOR ESTABLISHING PTC (ADDFNDUM 13)	(124+30) PROCEPURES FOR RETURN TO EARTH TARGETING (ADDENDUM 6)	(129+35) PROCEDURES FOR TERMINATING PTC ADDENDUM 14)	(129+40) PROCEDURES FOR IMU REALIGN TO REFSMMAT (ADDENDUM 2) (PB2)	BACKUP ALIGN NAV STARS NO. 14 (CANOPUS) NO. 12 (RIGEL) BACKUP ALIGN CHECK NAV STARS NO. 15 (SIRIUS)
USE ACS FOR DV LESS THAN AREQUAL TO LEFPS USE SPS FOR DV GREATER THAN 12FPS	TAIN DVS TO ZERO ************************************	(121+35) PROCEDURES FOM SM ACS PRODULS+ON MONITOR CHECKS (ADDENDUM 10)	(121+40) PROCFOURES FOR RCS THRUST SET P (ADDENDUM 12) (P41)	(122+00) SECOND TRANSEARTH MIDCOURSE (122+05)	PROCEDURES FOR SM RCS AND SPS PROPULSION MONITOR CHECKS (ADDENDUM 10)	(122+35) PROCEDURES FOR STAR/LUNAR HORIZON SIGHTINGS - 3 SETS OF 3 MARKS EACH (ADDENDUM 4)	RECOMMENDED NAV STARS NO• 2(DIPHDA) (NEAR HOR) NO• 2(DIPHDA) (NEAR HOR) NO• 1(ALPHERATZ)(NEAR HOR)

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(131+55) PROCEDURES FOR RETURN TO EARTH TARGETING (ADDEND''' 5)		(EI=15HRS)	(132.05)  BROCFDIDER FOR AND 110 TAK	OF CSM STATE VECTOR	(UPLINK INTO LM SLOTS ONLY)	GND UPLINK		POO LEB WHEN COMP 1,T-01,T (NO INTEG)	K K K K K K K K K K K K K K K K K K K	F 16 54 (R.RDOT.THETA)		Ora	KEY VAGE	F 04 06 (TRACK AXIS OPTION)	LOAD DOODS IN R2	F 06 19 (COMPUTED R.P.V.)	COPY R.P.	KEY VOAR		P00 KEY V37E00E	•	(140+25)	PROCEDURES FOR TERMINATING PTC	(ADDENDUM 14)	(140+30)
'129+55) PROCEDURES FOR GOC ALIGN TO ILU (ADDENDUM 3)	(ET-17HRS)		PROCEDURES FOR STAB/LUNAR HORIZON STGHTINGS - 2 SFTS		(ADDENDUM 4) P24)	PECOMMENDED NAV STARS NO. 2(DIPHDA)(NEAR HOR)	NO. 2(DIPHOA) (NEAR HOR)			(130+50)	PROCFOURES FOR STARZEARTH	HOPIZON SIGHTINGS - 6 SETS	OF 3 MARKS EACH	(ADDENDUM 4)	DECOMMENDED NAV STARS	JLUS)	NO. 22 (REGULUS)		PO 24:3F1C4)	ASS BY (ARCTURUS)	NO. 31 (AHCTURIS)			PROCEDURES FOR ESTABLISHING PFC (ADDENDUM 13)	

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NO. 31 (ARCTURUS) (NEAR HOR)	Đ	EARTH TARGETING (ADDENDUM 6)	PROCEDURES FOR IMU REALIGN TO REFSMALT		NO. 12(RIGEL) BACKUP ALIGN CHECK NAV STAR NO. 15(STRIUS)	PROCEDURES FOR GCC ALIGN TO IMU (ADDFNDUM 3)	IRS)	PROCEDURES FOR STAR	HCRIZON SIGHTINGS - 2 SETS OF 3 MARKS EACH (COMM FAILURE ONLY) (ACDENDUM 4)	RECOMMENDED NAV STARS NO. 22(REGULUS)(FAR HOR) NO. 26(SPICA) (NEAR HOR)	
	(14)+20)		(142+00)			(142,15)	(EI.4.5HRS)	(142+20)			(145+40)
PROCEDURES FOR IMU ALIGN	(ADDENÇUM 2) (PS2)	DACKUP ALIGN NAV STARS NO. 14 (CANOPUS)	NO. 12(RIGEL) BACKUP ALIGN CHECK NAV "T." NO. 15(SIRIUS)	(140+45) PROCEDUMES FOR GOC ALIGN TO INU (ADDENDUM 3)	(140+50) PROCEDURES FOR ESTABLISHING P+C (**ONLY IF COMM AVAI(ABLE**) (ADDENOUM 13)		THE FOLLOWING SIGHTING PERIOUS REDUINED ONLY FOR A COMM FAIL-RE	(EI=6HRS)	(140+50) PROCEDURES FOR STAR/EARTH HORIZON SIGHTINGS - 3 SETS OF 3 MARKS EACH	(ADDENDUM 4) (ADDENDUM 4)  PECOMMENDED NAV STARS	NO. 26 (SPICA) (NEAR HOR)

FINAL MISSION C-PRIME TRANSLUNAR, L'INAR PARKING ORBIT: AND TRANSEARTH TIMELINE -12 NOV 1968

PPOCEDURES FOR RETURN TO  FARTH TARGETING  (ADDENDUM 6)  ***********************************	PROPULSION SYSTEM SELECTION USE RCS FOR DV LESS THAN OR EQUAL TO 12FPS USF SPS FOR DV GREATER THAN 12FPS	TRIM DV TO ZERO ************************************	MONITOR CHECKS (ADDENDUM 10) PROCEDURES FOR RCS THRUST SET	(ADDENDUM 12) THIRD TRANSEARTH MIDCOURSE PROCEDURES FOR SM RCS AND SPS	PROPULSION MONITOR CHECKS (ADDFNDUM 10)	**************************************	HORIZON SIGHTINGS - 1 SET OF 3 MARKS (COMM FAILURE ONLY) (ADDENDUM 4) (ADDENDUM 4)
u v v v v v v v v v v v v v v v v v v v	OF NAU		REALIGN (P52)	æ		UPLINK OF ~SM P30 TARRET 10AD 1P27)	*

FINAL MISSION C-PRIME TRANSLUNAR, LINAR MARKING ORRITIAND TRANSEARTH TIMELINE -12 NOV 1968

RECOMMENDED NAV STAR NO. 33(ANTARES) (NEAR HOR)

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CM/SM SEPARATION (146+35)

(146+50)

ENTRY INTERFACE

FINAL MISSION C-PRIME TRANSLUNAR, LINAR PARKING ORRITIAND TRANSEARTH TIMELINE -12 NOV 1968

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	****	SPS THRUST SETUP (P40)
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ADDENDUM 1		RCS IMRUST SETUP
ASSUMED PRE-SEP SC CONFIGURATION	ZCE	ADDENDUM 13
ADDENDUM 2		ESTABLISH PTC
IMU ALIGNIOR REALIGN)	(P52)	ADDENOUM 14
ADDENDUM 3		
GDC ALIGN TO IMU		
ADDENDUM 4 STAR/FARTH(OR LUNAR) HORIZON		ORBITAL NAVIGATION PROCEDURES (P22) (PSEUDO LANDING SITE TRACKING)
_ <u></u>	(F24)	ADDENDUM 16
ADDENDUM S GROUND TRACK DETERMINATION	( jea,	ORBITAL NAVIGATION PROCENURES (P22) (LANDMARK TRACKING-NO AUTO OPTICS)
		ADDENUUM 17
AUDENDUM 6 RETURN TO EARTH TARGETING	(P37)	ORBITAL NAVIGATION PROCEDURES (P22) (LANDMARK TRACKING-WITH AUTO OPT)
ALDENDUM 7 GDC potet culos		ADDENDUM 18
		ORDEAL INITIALIZATION/VERIFICATION
ADTENDUM B	;	ADDENDUM 19
פאט הסר זאג	(P21)	HIGH BAIN ANT ACQUISITION
ADDERDUM 9 EXTERMAL OV TANGETING	(uEa/	
ADDENDUM 10 SM RCS AND SPS PROPULSION CHECKS	S S	

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ADDENDUM 1 ADSUMED PREINEP SC CONFIGURATION ************************************	ANDI-GO VC	•
	ELS LOGIC+OFF * ELS AUTO-MAN	
	CM RCS LOGIC-OFF CM PRPLNT DUMP-OFF	
	PPPLNT PURGED	<b></b> -
	O X L	4
	90° 0.00°	
	LV/SPS IND (2	<b>,</b> , ,
F PROCEDIJRE'S	GMB A CA	<b></b>
	FUNCTION-OFF	-
CDR**ASSUMED SWITCH POSITIONS**	MODE-STE	
		¥
FOAL SCALE-5/1		
SELECT	ACT STATE OF THE S	
FDAI SOURCE CMC		
5ET-60C	PL VENTAOFF	
MAN ATT (3) -RATF CMD		
CYCLE-0	IP TLMICA	~
ATT DB-MAX	AND AND	~
RATE-LOW		
THC LOCKED	L 8 CB CLOSED	•
OCKEU	BAT A,8AT 8(2) -OPE	€0
CON TERIORE	SECS ARM (BOTH) = 0	
SON TO THE SECOND SECON	S LOGIC (BOTH)	€0
THE CHOIL CHOIL THE	PL VENT FLT/PL=0PE	<b>C</b>
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ACIDE AND ACIDE	ACS LOGIC (S) -OPEN	<b>6</b> 0
MOUE (3) SHA	<b>►</b> O <b>∀</b>	•
SPS THRUST DIRECTINOREAL DO THRUST AND SPE	FLOAT BAG(ALL) - VENT	<b>@</b> (
IVC (2) -ALITO	ALDE (EFCE) DIEDO DOUG	<b>D</b> (
	TOPE STANDARD TO	<b>3</b> 0
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	04 46 (DAP CONFI LOAD 11103 111111 PRO 111111 PRO +XXXXX +00000 PRC +XXXXX	<b>4</b> 7
FOS PWR-OFF  TVC SERVO PWR(2) - OFF FOAI/GDI PWR-BOTH LOGIC PWR-BOTH SCS ELEC PWR-GOC/ECA SIG CONDR/DR BTAS(2) - AC1 RMAG PWR(dOTH) - ON 7 LMP**ASSUMFO SWITCH POSITIONS**	AC1 IE(2) - OFF R-AC1 US-MNB G/N CB(10) - CLOSED FYRO(2) - OPEN NT OMNI- REGD OMNI ANT REGD OMNI ANT OHM MODE VOICF-VOTCE OHM MODE PCM-PCM	S-BAND AUX TV-OFF S-BAND AUX TV-OFF UP TLM DATA-DATA  IIP TLM CMD-NORM  G/N PWR.OPTICS-DN  G/N PWR IMU-ON  100  OPT MODE-MAN  122  OPT ZERO-ZERO

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UPLINKED PREFERRED REFSMMATAND GO TO F 06 22 DISPLAY OR LOAD 00003 FOR REALIGN TO

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F 01 71 (MARKED STAR CODE)	F 01 70 (STAR CODE) CHECK SECOND STAR CODE	DRO PRO DE CMC DE GENERAL DE CMC DE COMPANA DE CMC DE COMPANA DE CMC DE	MONITOR OPTICS DRIVE TO STAR TWO IDENTIFY STAR TWO	OPT MODE-MAN F 51 88 (PLEASE MARK) MARK ON STAR TWO WITH SXT	F SO 25 (00016, TFRMINATE MARK SEQ)	F 01 71 (MARKED STAR CODE) PR <sub>0</sub>	F 06 05 (ANGLE DIFF) IF GREATER THAN .OS DEG.KEY	V32F.GO TO F SO 25(00014) And repeat alignment	IF .OS DEG OR LESS.PRO F 06 93 (GYRO TORQ ANGLES) LMF RECORD GYRO TORQ ANGLES	VOICE TO GND PAO IF GYRO TURO A (00014, PERFORM FI	IF REPEAT ALIGNMENT REGULRED. PRO.RETURN TO F 50 25 (00015). VERIFY STAR TO AND REPEAT	MARK PROCESS ON TWO STARS IF ONLY ALIGN CHECK REQUIRED.	KEY ENTER, GO TO F OI 70. AND VERIFY SXT DRIVES TO THIRD STAR
REFSMMAT AND AD TO	PRO PROPOSED ALIGNMENT)	RIFY NO GMHL LOCK Roposed Alignment O/monitor Coarse	F 50 75 (00015, PERFORM STAR ACQ) RHC-ARMED PRO	IF TWO STARS NOT AVALLABLE.A F US 09 (CODE 405) ALARM WILL OCCUR. COR MANUALLY	MANEUVER SC TO ACQUINE BACKUP ALIGN NAV STARS IN SCT FAV.	CMP KEY V3ZE+RFTLIRN TO F 50 75 (00015) NISPLAY AND PRO	ILABLE AT	F 01 70 (STAR CODE) CHECK FIRST STAR CODE	OR LOAD THIRD STAR COURT IF DOING ALIGN CHECK OPT MODE-CMC	PRO 06 92 (SHAFT, TRUN, BLANK) MONITOR OPTICS DRIVE TO	STAR ONE (IF VERIFYING SXT DRIVE 1) THIRD STABLES VATEORS)	IDENTIFY STAR ONE OPT MODE NAME OF STAR ONE OPT MODE NAME OF STAR OF S	205

FINAL MISSION C.PRIME PROCEDURFS DATED-12 NOV 1968

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ADDENDUM 3 GDC ALIGN TO IMU ************************************	F P P P P P P P P P P P P P P P P P P P	**PROCEDURES FOR GDC ALIAN TO IMU**	COR ATT DB-MIN  BMAG WODE (3) -ATTI/RATE2  BMAN ATT (3) -RATE CMD  SC CONT-SCS  FDAI SELECT-1  FDAI SELECT-1  FDAI SCURCE-ATT SET  ATT SET-IMU  KEY VIGNEDE  16 20 (CURRENT R.P.Y)  NULL ATT FPROR NEEOLES  ON FDAI 1 WITH  ATT SET THUMBWHEELS  (AVOID FALSE NULL RY  COMPARING ATT SET IND  WITH NZO)  ATT SET-GDC  DEPRESS GDC ALIGN BUTTON  FDAI SELECT 1/2  ATT OR-MAX	50
CALLED AFTER THIRD STAR "HK; OPT MODE=MAN OPT YET TRUN TO LESS THAN 5 DER OPT ZERO-ZERO RETICLE BRT TW-MIN BRTNES«	@ M ← Ø K O O		C C C C C C C C C C C C C C C C C C C	11

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ADDENDUM 4 STAR/EARTH(OR LUNAR) HOR (P23) (OR LMK SIGHTINGS)	S E PROCEDURES	***PROCEDURES FOR CISLUNAR NAV***	LMP G/N PWR-ACI TLM INPUTS PCM-LOW UP TLM CMD-RESET THEN NORM TAPE RCDR RCD-RCD TAPE RCDR RCD-RCD TAPE RCDR RCD-RCD TAPE RCDR RCD-RCD TAPE MOTION TB-BP LEG G/N PWR-OPTICS-ON (OPTICS PWR MUST BE ON AT LEAST 30 MIN PRIOR TO TAKING MARKS) OPT ZERO-ZERO (155EC) OPT ZERO-ZERO (155EC) OPT TELTRUN-SLAVE TO SXT COND LAMPS-ON OPT MODE-MAN RETCL BRT TW-ADJUST COR SC CONT-CMC CMC MODE-MAN BMAG MODE (3)-RATE2 MAN ATT (3)-RATE2
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SUPERIMPOSE SXT STAR LOS ON SXT LMK LOS WITH THE OPTICS HAND CONTROLLER MARK WHEN THE TWO LINES OF	SIGHT ARE SUPERIMPOSED F 06 87 (BLANK, TRUN ANGLE BIAS, BLANK) COPY TRUN ANGLE BIAS	F 59 BB (PERFORM CALIBRATION MARK) SUPERIMPOSE SXT STAR LOS ON THE SXT LMK LOS MARK WHEN THE TWO LINES OF SIGHT ARE SUPERIMPOSED	F 06 87 (BLANK, TRUN ANGLE BIAS, BLANK)	IF THE TWO TRUN ANGLE BIAS MEASUREMENTS ARE NOT WITHIN *003 DEG. KEY V3ZE AND REPEAT THE	CALIBRATION PROCESS IF ANY TWO TRUN ANGLE BIAS MEASUREMENTS ARE WITHIN .003 DEG. PROCINCORPOBATE TRUN BIAS)	F 51 BB (PLEASE MARK)  KEY VOAE (PERFORM LMK(HOR)  ACQUISITION)  F 50 18 (COMMANDED R,P,Y ANGLES)		O6 18 (COMMANDED R.P.Y ANGLES) COR MONITOR AUTOMATIC MANEUVER FOR GIMBAL LOCK LEB MONITOR AUTO MANEUVER, VERIFY LMK/HOR IN SXT FIELD OF VIEW
KEY V37E23E . 05 70 (MEASUREMENT INENTIFICATION) LOAD	XX 0 XX X 0 X 0 0 X 0	FOR LMK MARKS, RCHUUND FOR LMK MARKS, R3=00000 AND LOAD LMK DATA IN R2 (ASSUMED ALL KNOWN LMKS,	0 25	" 59 RB (PENFORM CALIBRATION MARK) CALIBRATION OF OPTICS TRUNION REQUIRED AT THE REGINNING OF EACH SIGHTING PERTOD.	IF OPTICS CALIMRATION ALREADY PERFORMED. OPT MODE-CMC OPT ZERO-OFF KEY ENTER AND GO TO	16 92 DISPLAY 1F OPTICS CALIHRATION REGUIRED OPT MODE=MAN RHC 2-ARMED	MANELVER SC TO ACQUIRF A «TAD WITH THE S'T LMK LOS COR CMC MODE=FREE	LER FIX STAR IN SXT LMK LOS F+ELN OF VIEW WITH MIN IMP CON+PGI OPT ZERO-OFF OPT COUPLING-RSLV OPT SPEED-'OW

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DISPLAY, AND TEPEAT MARK F 37 88	A STAP/LMK(HOR) SET 23E.RETURN TC 5 70 DISPLAY RROCESS	WHEN SELECTING A STAP/LMK(HOR) SE KFY 23E, PETURN T F 05 70 DISPLAY REQUIRED ATT MAN TAKF THREE MARKS WHEN TERMINATING SIGHTING PERIOD	POO KEY OOF OPT MODE-MAN ORIJE TRUN LESS THAN S DEG OPT ZERO-ZERO	HETICLE BRT TWEMIN BRINESS COR CMC MODELAU70 LMP IF TAFE RCOR O WRATING TAPE RCOR FWD+OFF (GTR) TAPE MOTION TR-GRAY	(ASSUME GND PLAYBACK AT NEXT OPPORTUNITY)	
F 50 18 (COMMANDED R.P.Y ANGLES) OPT MODE-CMC	06 92 (COMMANDED OPTICS ANGLES) MONITOP OPTICS DRIVE TO SELECTED STAR	F 51 RB (PLEASE MARK ON STAR-, MK/WOR) COR CMC MODE-FREE LER MANEUVFR SC WITH MIN TMP CONTROLLEH TO POSITION LMK (HQR) IN SXT AT SIE STELLAR POTNT FIX LMK (HQR) IN SXT OPT COUPLING-RSLV	OPT SPEED-LO Maneuver Optics to SuderiPoge Star O., Lmk/Hor Mark On Star-Lmk/Hor	F 50 75 (00016.TERMINATE MARK SEG.) PRO F 05 71 (MEASUREMENT IDENTIFICATION) VERIFY MARKED DATA PRO	F 06 49 (DR.DV.BLANK)  **HOLD AT THIS DISPLAY FOR  ZO SECONDS**  KEY VOINOIE 362E  CCPY TRUN (OCTAL) IN RI  **HOLD AT THIS DISPLAY FOR  KFY BELEASE	F 06 49 (DE.DV-BLANK) PRO IF DR AND DV ACCEPTABLE OR IF NOT ACCEPTABLE; REY V32E, RETURN TO F 05 70

FINAL MISSION C.PP'ME PROCEDURES DATEN-12 NOV 1964

	ADDENDOM DARTH TARGETING (PW)	**PROCEDURES FOR RETURN TO EARTH TARGETING *	) (4)	33 (TIG OF MIDCOURSE)	LOAD TIG (GET) OF MIDCOURSE PRO	60 (BLANK, VPRED, GAMMA EI)	LOAD ZERO IN RU(VERED) AND	040	KEY CIGNURE TONITOR ITERATION	KEY BELEASE EHEN KEY REL LIBON	613 OCCUR, KEY V32E, GO TO	F OF 33 DISPLAY, AND ADJUST	INPUT PARAMETERS	61 (IMPACT LAT.IMPACT LONG.BLANK)	ACCEPTABLE - KEY V32F - GO TO	F 06 33 DISPLAY, AND ADJUST	INPLI PARAMETERS		39 (DELTA T TRANSFER)	COPY DELTA T TRANSFER (GET)	AND IF NOT ACCEPTABLE TRY	VAZE, GO TO F OB 33 DISPLAY,	DAD ADCOM TATOM DENEMBERS DAD 18 ACCEPTED F	60 (BLANK, VPRED, GAMMA EI)
•	4 K +	•	(	9		90								90					90					90
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			000	Š																				
· · · · · · · · · · · · · · · · · · ·	DIENDOM DEFERMINATION (PC))		<b>Q.</b> ◀	(Z	PROCEDURES		KEY V37621	(00002,VE	NI 10000 C	CAG.	LOAD NOMINAL TIME OF	LOII RECEIVED FROM GND	020	CLAT+LONG+ALT)	IF A LOMIN INCREMENT FROM THE	LOADED TIME IS DESIRED.	KEY VAZE AND RETURN TO	F 06 34 DISPLAY (TIME IS	AUTOMATICALLY INCREMENTER)	IF PROGRAM TERMINATION	IS DESIRED.	2.	7 × 7 × 7 × 7 × 7 × 7 × 7 × 7 × 7 × 7 ×	, -
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VERIFY RZ=00001 FOR SPS OR
                                 LOAN R2=00002 FOR ACS
PRo
                                                           (TIG OF MIDCOURSE)
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COPY VPRED AND GAMMA FI AND
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                                                       PRO IF ACCEPTANLE (DELTA VXVVVVVZ-LOCAL VERT)
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                           GO TO F 06 33 DISPLAY, AND ADJUST INPUT PARAMETERS
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FINAL MISSION C.PRIME PROCEDURES DATEN-IZ NOV 1968

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                                                                                                                                                                                     RATE-LOW
RMAG MODE (3) -ATTI/RATE2
MAN ATT(3) -RATE CMD
SC COM1-SCS
KEY VOGNZOE
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FUAI SCURCE-ATT SET
FUAI SCURCE-ATT SET
ATT SET-GOC
NULL ATT ERROR NEEDLES
ON FDAI 1 WITH
ATT SET THUMWHEELS
(AVOID FALSE NULL BY
COMPARING ATT SET IND
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           ADDENDUM 7
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FINAL MISSION C-PRIME PROCEDURES DATEN-12 NOV 1969

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LMP RECORD DRIFT DATA AND VOICE TO GND COR ATT DB-MAX

FOAT SEL-1/2

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ADDENDUM B GND UPLINK ************************************

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FINAL MISSION C.PRIME PROCEDURES DATED-12 NOV 196A

ADDENDUM 10	**PROC FOR SPS MONITORING CHECK**
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	+53 TO +75 DEG
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	ITY TSOEG. SPS LINE 1785-0FF
	N28
Q.	SPS HE/N2 PRESS IND
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PROCEDURES	<b>t</b>
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**PROC FOR SA ACS MONITOR CHECK **	
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FINAL MISSION C-PRIME PROCEDURFS DATEN-12 NOV 1968

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EMS FUNCTION-DV TEST MONITOR SPS LT-ON MONITOR DV COUNTDOWN MONITOR SPS LT-OFF CONFIRM DV = -20.8+/420.7 EMS MODE-STBY EMS FUNCTION-DV SET LOAD BURN VC EMS FUNCTION-DV		COMMANDED R.P.Y) PRO (COMMANDED R.P.Y) MONITOR ATT MANEUVER FOR GMBL LOCK (COMMANDED R.P.Y) VERIFY R.P.Y AGREE WITH BURN PAD R.P.Y AGREE WITH BURN KEY ENTER	KEY V37E40E  (COMMANDED R.P.Y)  VERIFY R.P.Y AGREE WITH PAD  VALUES WITHIN 5 DEG  ALIGN SC IN ROLL TO PAD VALUE  PRO (COMMANDED R.P.Y)
	06 22 00 20 00R	50 18 06 18 06 18 50 18	
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			4
######################################	Y E E E E E E E E E E E E E E E E E E E	LEB KEY V37E00E CDR CONFIRM GND PIE& BIAS CHECK **PROCEDURES FOR EMS TEST AND &/U4** CDR CB.EMS(2)/CLOSED SC CONY=CMC CMC MODE=FREE EMS MODE=STBY EMS FUNCTION=OFF	EMS MODE-AUTO VERTEY DV LESS THAN 3.47PS PER 100SEC CMC MODE-AUTO EMS FUNCTION-OFF EMS MODE-STBY EMS MODE-STBY EMS MODE-AUTO LOAD DV # 1586.8

FINAL MISSION C.PRIME PROCEDURES DATED-12 NOV 1968

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FINAL MISSION C-PRIME PROCEDURES DATEN-12 NOV 1968

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(LESS THAN 2FPS PER 5 SEC)  (IF RFQUIRED)  (IF RFQUIRED)  (MP PRO  0 06 40 (TFC.VG.DVM)  0 06 40 (TFC.VG.DVM)  10 06 40 (TFC.VG.DVM)  COM MONITOR SPS LT.ON/START  MDC ET COUNT UP  TERMINATE ULLAGE 1 SEC  AFTER SPS IGNITION  CONFIRM FOAI 1 RATES LESS THAN  10 DEG/SEC  MONITOR PC GAUGE 95.105 PSIA  CMP VERIFY VG/DECR AND DVM/INCR  MONITOR FOAI 2 RATES LESS THAN	10 DEG/SEC LMP HONITOR VALVES OPEN AND SPS GAUGES CDR MONITOR SPS ENGINE CUTOFF MONITOR SPS 'IT-OFF F 16 40 (TFC,VG,DVH) (COPY VG AND DVM) CUIOFF+1SEC DV THRUST (BOTA)-OFF MONITOR PC GAUGE ZERO LMP MONITOR VALVES CLOSED SPS HE VLV TB (BUTH)-BP FLT RCDR-OFF (CTR) CDR CB SPS P2-CLOSED CB SPS P2-CLOSED SPS GMBL MTRS (4)-OVF (SEQUENTIALLY-P2,V)-P1) TVC SERVO PWR (BOTH)-OFF	CMP PRO F 16 85 (VG-RODY) CDR NULL VG RESIDUALS AS RESD THC-LOCKED	<b>5</b> 1
ATT SET TW (AVOID FALSE NULL BY COMPARTING ATT SET IND WITH N20) ATT SET-GOC DEPRESS GOC ALTGN PB FDAI SELECT-1/2 CMP KEY RELEASE F 50 25 (00204, GMBL DRIVE TEST) CMP PRO COR MONITOR GMBL DRIVE SEG AND TRIM O6 40 (TFI-VG.DVM) VERIFY SPS GPI TRIM AGREEC WITH PAD WITHIN 0.5 DEG	LMP NONESS BUS-MNA TAPE RCDR FWO-OFF (CTR) TLM INPUTS PCM-HIGH UP TLM CMD-RESET THEN NOR; TAPE RCDR PCM-PCM/ANLR TAPE RCDR PCM-PCM/ANLR TAPE RCDR PCM-PCM/ANLR TAPE RCDR PCM-FWO TAPE RCDR FWO-FWO THRUST (BOTH) -NORMAL THC-ARMED RHC-ARMED	-35SEC DSKY BLANKS -30 06 40 (TFI.VG.DVM) COR EMS MODE-AUTO -25 CMP CK DVM FOR HI PIPA BIAS	33

FINAL MISSION C.PRIME PROCEDURES DATED-12 NOV 1968

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FINAL MISSION C-PRIME PROCEDURES DATEN-12 NOV 1968

LOAD PAD SHAFT ANGLF(+XXX,XX) IN R1 AND PAD TRUNNION ANGLE (+XX,XXX) IN R2 41 BB (SHAFT,TRUNNION,RLANK)	MONITOR OOT DRIVE TO PAD VALUES F 50 18 (COMMANDED R.P.V) VERIFY PAD STAR IN SXT	OPT MODE=MAN DRIVE TRUN LESS THO OPT ZERO=ZERO HETICLE RRT TW=MIN MOVE TO CENTER SEAT KEY ENTER	Ö	+*ALIGN GDC TO IMU++ CDR FDAI SELECT 1 FDAI SOURCE-ATT SET ATT SET-IMU CMP KEY VIGNZOE 16 ZO (R,P,Y) COR NULL FDAI NEEDLES WITH ATT SET TW (AVOID FALSE NULL BY COMPARING ATT SET IND WITH NZO)	ATT SET-GDC DEPRESS GDC ALIGN PR FOAI SELECT-1/2 CMP KEY RELEASE 38
F 06 22 (FINAL GMBL ANGLES) LOAD BURN R.P.Y GMBL ANGLES KEY V62E COR RMAG MODE(3)-RATE2	PRO (COMMANDED R.P.Y) PRO (COMMANDED R.P.Y)	COR MONITOR ATT MANFUVER FOR GMBL LOCK F 50 18 (COMMANDED R.P.Y) VFQIFY R.P.Y AGREE WITH JAO R.P.Y WITHIN 5 DFG LEB KEY ENTER	*PROC FOR RCS THE LEB KEY V37E41E 18 (COMMANDED R COM ALIGN SC IN F LEB PRO	O6 18 (COMMANDED R.", Y) CDR MONITOR BURN ATT TRIM F 50 18 (COMMANDED R.P.Y) VERIFY R.P.Y AGREE WITH P.D VALUES WITHIN 5 DEG LEB OPT MODE_MAN G/N PWR OPTICS_ON OPT ZERO-ZERO(15SEC) OPT ZERO-ZERO(15SEC) OPT ZERO-ZERO(15SEC)	OPT ZERO-OFF OPT MODE-CMC KEY V41N91E F 21 92 (BLANK,BLANK) 37

FINAL MISSION C-PRIME PROCEDURES DATEN-12 NOV 1968

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IS (VG-BODY)

EMS MODE-AUTO

P KEY N40E

O (TFI.VG.DVM)

MONITOR DVM FOR HT PIPA BTAS

(LESS THAN 2FPS PER 5SEC,

KEY RELEASE
           UP TLM CMD_RESET THEN NOR:
TAPE RCDR PCM-PCM/ANLG
TAPE RCDR PCD-RCD
TAPE RCOR FW)-FWD
TAPE MOTION TB-RP
DSKY BLANKS
                                                                                                                                                                                                                                        SET MUC ET COUNTING UP ANNULL VG AS REQUIRED
LMP TLM INPUTS PCM_HIGH
                                                                                                                                                                                                                                                                                                                                      **CALL IDLING PROGRAM**
                                                                                                                                                                                                                                                                                RHC-LOCKED
CMP PHO
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                                                                                                                                                                                                             THC-ARMED
                                                                                                                                                                                                                          F 16 A5 (VG-BODY)
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                                                                                                                                          16 40
                                                                                                                             -25 SEC CMP
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KEY VOKE (TRANSFER POST BIRN SV FROM CJM SLOTS TO LM SLOTS) LMP TAPE ACOR FWO-OFF(CTR) TAPE MOTION TB-GRAY (ASSUME GND PLAYBACK AT NEXT OPPORTUNITY) EMS MODE-STBY KEY OOF COR

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FINAL MISSION C.PRIME PROCEDURES DITER-12 NOV 196A

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EMS FUNCTION-OFF

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OR USE PREVIOUS PTC ANGLES IF STILL VALID.COORDINATE PT GMBL ANGLE UPDATES WITH GND	CMP KEY V49E 06 22 (FINAL GMBL ANGLES) LOAD PTC GMBL ANGLES KEY V62E RHC-ARMED	COMMANDED R. COMMANDED R. COMMANDED R. COMMANDED R.	MANUEL LO MANUEL	MACTON TO THE COLUMN TO THE CO	Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z
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CDR ATT DB\_MIN

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PROCEDURES

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(DAP CONFIGURATION)
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11111

CMP KEY V48E

F 04 46

CSM AND LM MT)

F 06 47

PRO F 06 48 (SPS GMBL TRIM)

C W Z W J

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LMP RECEIVE PTC R.P.Y FROM GND

ADDENDUM 13 ESTABLISH PTC

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FINAL MISSION C-PRIME PROCEDURES DATEN-12 NOV 1968

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FINAL MISSION C-PRIME PROCEDURES DATEN-12 NOV 1968

SITE LEB OPT MODE=MAN F 31 88 (PLEASE MARK) MAKE 5 MARKS=20 SEC APART POSITION TRUNNION TO 5 DEG	COR STOP PITCH RATE  LEB PRO  F 05 71 (LANDING SITE DATA)  VERIFY R2  LEB PRO  F 06 49 (UELTA R.DELTA V.BLANK)  "ECOPD VALUES  **HOLD AT THIS DISPLAY FOR	F GS 89 (LAT.LUNG/2.ALT) RECORD VALUES **HOLD AT THIS DISPLAY FOR ZO SECONDS** LEB KEY V37E00E	94
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FINAL MISSION C.PRIME PROCEDURES DATED-12 NOV 1968

POSITION TRUNNION TO 5 DEG OPT 7ERO-ZERO CDR STOP PITCH RATE MAN ATT(P)-ACCEL CMD PITCH SC UP TO 5 DEG PITCH AND 0 DEG ROLL W.R.T.	LOCAL HORIZONTAL  HAN 677(P)-MIN IMP  INITIATE ~ REITAL PITCH RATE  FOR WFXT LANDWARK SIGHTING  LEH PRO  F 05 71 (LANDWARK DATA)  VERIFY R2  LEB PRO  F 06 49 (DELTA R.DELTA V.BLANK)  HECORD VALUES	FOR SECONDS **  VALUES  THIS DIS _AV FOR SECONDS **  FOR FOR SECONDS **	wo.
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FINAL MISSION C-PRIME PROCEDURES DATEN-12 NOV 1968

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######################################			LEB KEY V3722E F 06 45 (BLANK, BLANK, MAX MGA)	F 05 70 (LANDMARK CODE) LOAD 10000 IN R2 LEB PRO	LEB	PRO O6 92 (SHAFT, TRUN, BLANK) MONITOR OPT DRIVE TO LANDWARK IDENTIFY IDENTIFICATION PAINTS IF VISABLE LEB.COR WHEN LANDMARK IDENTIFIED— START 0,3 DEG/SEC	6*

FINAL MISSION C.PRIME PROCEDURES DATED-12 NOV 1964

CDR SET ORDEAL ON FDAI FDAI/GPI PWR-80TH LOGIC 2/3 PWR-ON CMC ATT-IMU FDAI SELECT-1/2 FDAI 1-0PB RATE

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C R E W PROCEDURES

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MODE-HOLD/FAST F 04 06 (00002,00001)

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KEY V83E F 16 54 (R,RDOT,THETA)

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FINAL MISSION C-PRIME PROCEDURFS DATEN-12 NOV 1968

LEB KEY V37E00E

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GMBL COORDINATES
CMP PRO
LMP HI GAIN ANT POSITION
SET ANT PITCH TO RHO
SET ANT YAW TO GAMMA
S-BAND ANT OWNI-HI GAIN
VERIFY HI GAIN S-BAND
ANT IND GREATER THAN
HALF SCALE
HI GAIN ANT TRACKAUTO OR REACO AS REQUIRED
HI GAIN ANT BEAM-AS REQUIRED LMP HI GAIN ANT TRACK-MAN HI GAIN ANT BEAM-WIDE F 06 51 (RHO, GAMMA, BLANK) LMP RECORD HI GAIN ANT CMP KEY V64E

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ADDENDUM 19 HIGH GAIN ANT ACQUISITION

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FINAL MISSION C-PRIME PROCEDURES DATED-12 NOV 1968

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